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ENVIRONMENTAL EFFECTS ON ADVANCED COMPOSITES

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INTRODUCTION

This is the final report for NASA-Langley Grant # NSG.1531.

This report is divided into four parts:

Part 1 - Mechanical Property Degradation and Chemical Interactions
in a Borsic/Titanium Composite.

Part 2 - Interfacial Coatings for SiC/Ti Composites

Part 3 - Solutions for Diffusion in Single-, Two-, and Three-
Phase Binary Alloy Systems.

Part 4 - Moisture Effects on Graphite/Polyimide Composites.

The first two parts deal with the development of titanium matrix composites for elevated temperature applications. Part 3 represents progress toward the ultimate goal of treating interactions in multiphase multicomponent systems. Part 4 represents the progress in support of NASA's CASTS program.

The research in this grant resulted in several publications. These are listed in the References Section.

PART 1 - MECHANICAL PROPERTY DEGRADATION AND CHEMICAL
INTERACTIONS IN A BORSIC/TITANIUM COMPOSITE

OBJECTIVE

Titanium-alloy matrix composites have the potential for high specific strength and stiffness in high temperature applications. These fiber-reinforced materials have unique properties which make them attractive for such applications as advanced engine components and a variety of aerospace structural components. However, the use of titanium composites has been limited because of poor as-fabricated properties and/or degradation of mechanical properties in high-temperature environments because of fiber/matrix interface interactions. The objective of this study is to identify the mechanisms of mechanical property degradation in a Borsic (boron coated with silicon carbide) fiber reinforced Ti-3Al-2.5V composite exposed to elevated temperatures.

APPROACH

Samples containing 0.45 volume fraction of fibers were exposed, in vacuum, to temperatures from 700 K to 1255 K for times up to 240 hours. Room temperature tensile properties of unidirectional material were determined in both the longitudinal and transverse directions, before and after high-temperature exposure. Electron microprobe analysis, scanning electron microscopy, and X-ray diffraction were used to determine the compounds formed and the extent of interaction between the boron, SiC coating, and matrix materials. The results of this study are summarized below, and the details are presented in reference 1.

RESULTS AND DISCUSSION

High temperature exposure had no effect on the longitudinal modulus of the composite whereas the transverse modulus for the thermally exposed specimens was about 20 percent greater than that for the as-fabricated specimens. The strength of the composite did not depend to any great extent on the exposure time for temperatures of 922 K or less, but was degraded significantly by the longer exposure times at temperatures of 1033 K and greater. For all exposure times, the strength after exposures at 811 K and 922 K was greater than that after 700 K exposures. No quantitative correlation between strength and reaction-zone thickness was found.

Electron microprobe, X-ray diffraction, and reaction-zone thickness data suggest a two-stage reaction process. The first stage consisted of the simultaneous interdiffusion of the Si, C, and Ti resulting in the depletion of the SiC coating and formation of titanium silicides (Ti_5Si_3 , TiSi , TiSi_2). The second stage resulted in significant titanium boride (TiB , TiB_2 ; Ti_3B_4) formation as well as a higher rate of formation of TiSi_2 . The strength of the composite was degraded before the formation of any identifiable boride compounds. The aluminum in the matrix did not take part in the reactions and was rejected from the reaction region, whereas the vanadium remained throughout the reaction zone.

PUBLICATIONS AND TALKS

The research on this study resulted in the following publication: Mechanical Property Degradation and Chemical Interactions in a Borsic/Titanium Composite, Proceedings of the 24th National SAMPE Symposium and Exhibition, San Francisco, CA., 1979.

PART 2 - INTERFACIAL COATINGS FOR SiC/Ti COMPOSITES

OBJECTIVE

The concept of interfacial coatings, to provide reduced chemical interaction between the SiC fiber and titanium matrix, is the subject of this study. The objectives are: (1) to investigate the effectiveness of interfacial coatings, in reducing the reactivity between fiber and matrix, (2) to identify the intermetallic compounds responsible for the degradation of the as-fabricated and heat-treated composite mechanical properties, (3) to determine any modification of the reaction zone compounds in the composites with interfacial coatings compared to the uncoated titanium alloys, (4) to provide a mechanical and chemical characterization of the SiC fiber for a better understanding of its role in the degradation of the composite mechanical properties.

APPROACH

The interfacial coatings selected for examination were aluminum, molybdenum and vanadium. These coatings were chosen because they were constituents of the alpha-beta alloy, Ti (6Al-4V), and certain beta alloys of titanium, which are known to have reduced reactivity with SiC compared to unalloyed titanium. The interfacial coatings were radio frequency (RF) sputtered onto Ti (A55) sheets and diffusion bonded (by DWA Composite Specialties in Chatsworth, CA) into unidirectional, SiC reinforced composite panels. Additional panels of SiC reinforced Ti(6Al-4V), Ti(3Al-2.5V) and Ti(A55), with no interfacial coatings, were fabricated

(by TRW Incorporated in Cleveland, OH) to provide a comparative evaluation of the mechanical properties, reaction compounds and reaction zone thicknesses with the coated composite systems. The results of this study form the basis of Mr. Larry House's M.S. Thesis (ref. 2). The author of this report, Dr. Unnam, worked closely with Mr. House on this study.

RESULTS AND DISCUSSION

An interest in SiC reinforced titanium composites has been generated by the strength loss at elevated temperatures (near 1000^oF) of boron and Borsic fibers, and a need for stronger matrices with higher temperature capabilities (up to 1400^oF) than can be provided by aluminum and resin matrix composites. Also, the reaction kinetics are reported to be slower for the SiC/Ti system than for B/Ti, thereby reducing the deleterious effects associated with fiber-matrix interaction. However, the poor as-fabricated mechanical properties of the SiC/Ti composites have precluded extensive utilization of this composite system. The best results were achieved with SiC/Ti (6Al-4V) system which yielded only 75% of the rule-of-mixture fracture strength.

The degradation of the tensile strength is largely due to the formation of a brittle phase boundary at the fiber/matrix interface. The strength loss is attributed to the stress intensification caused by defects or cracks that develop because of the low strengths and strain-to-failure of the reaction compounds.

Two, crucial, technical issues determining the utility of the SiC-Ti composites remain to be resolved. These are: (1) whether the fiber-matrix reaction kinetics during fabrication can be sufficiently reduced to prevent forming a reaction zone thickness greater than the critical thickness, and (2) if so, whether the composite degradation at the expected service temperature will be small.

Although the chemical compatibility between SiC and Ti has not proven satisfactory, several approaches to reduce the fiber-matrix reaction have been suggested in the literature. Among these were: (1) optimization of the consolidation process, (2) development of a low-reactivity matrix, (3) development of protective interfacial coatings. The first two approaches have been given considerable attention with only limited success. The development of protective interfacial coatings is the approach taken in this study.

Reaction Zone Thicknesses:

A comparison of reaction zone thicknesses for each composite system, after the 1600 and 1800^oF exposures, is shown in figure 1. The reaction zone thicknesses of the SiC-Ti (A55)/Al, SiC-Ti(3Al-2.5V) and SiC-Ti(6Al-4V), after the 1600^oF-25 hour exposure, were comparable. However, the composites with molybdenum and vanadium interfacial coatings have reaction thicknesses, after the 1600^oF exposure, comparable to the thicknesses in the 1800^oF exposed SiC-Ti(3Al-2.5V) and SiC-Ti(6Al-4V) specimens. Indeed, the composites

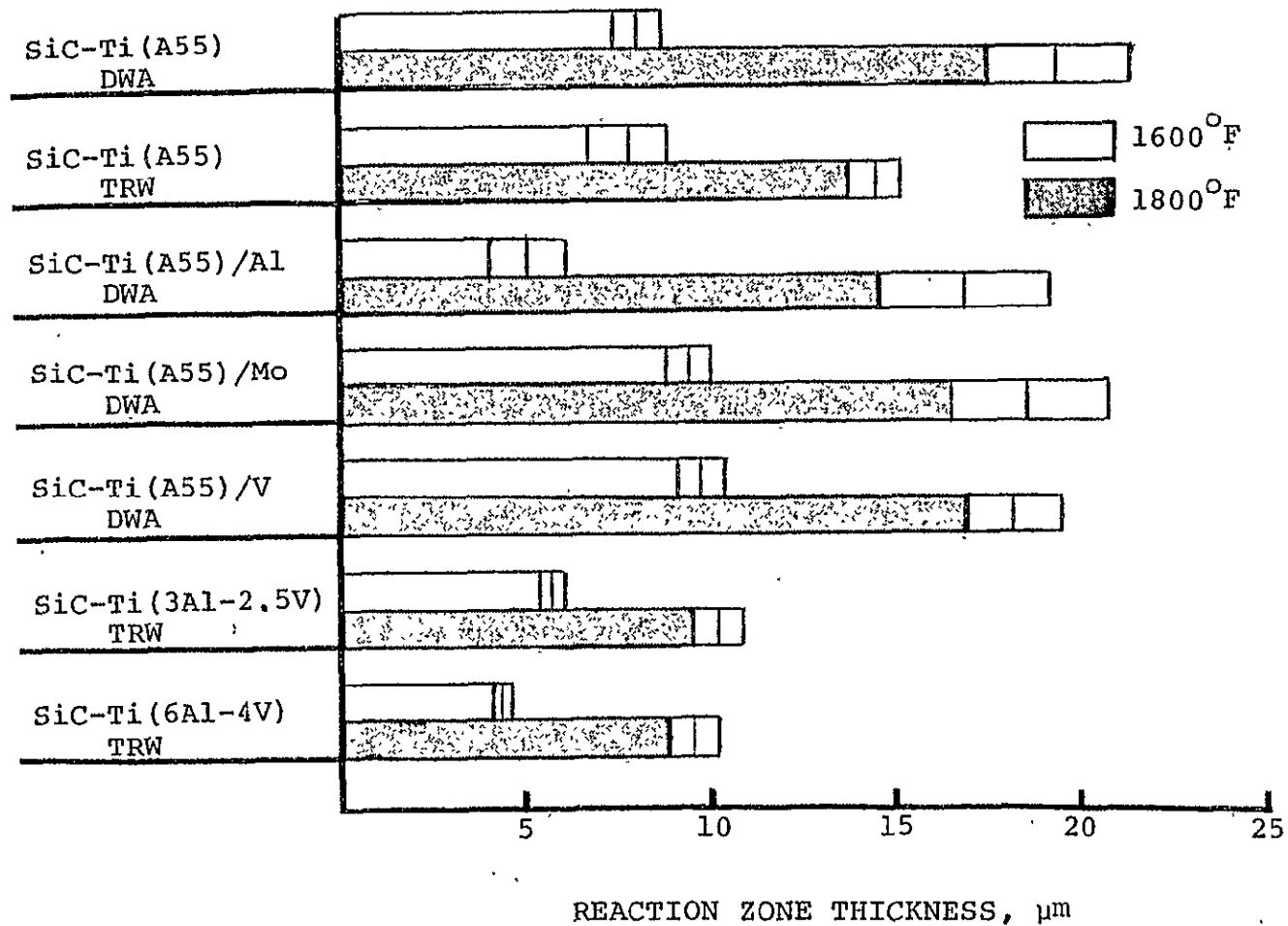


Figure 1. A comparison of the reaction zone thicknesses for the various composite systems after the 25 hour exposure at 1600 and 1800°F.

with molybdenum and vanadium coatings appear to have reacted, at 1600°F, more extensively than the composites with the uncoated Ti (A55) matrix. After the 1800°F exposure, all the DWA composites, with and without interfacial coatings, had similar reaction zone thicknesses. This should be expected because the homogenization of the coatings throughout the matrix leaves an insufficient amount of the coating to affect the fiber-matrix reaction. The larger reaction zone thickness for the 1800°F exposed (DWA) SiC-Ti (A55) specimen, compared to the (TRW) SiC-(A55) composite, was due to larger TiC precipitates in the outer boundary of the reaction zone in the DWA specimen.

The reaction zone thickness data for the 1600°F exposure suggest that a binary Ti-Al alloy matrix may be less reactive than the Ti (3Al-2.5V) and Ti (6Al-4V) matrices. This conclusion is made in view of the enhanced reactivity of the V coated composite, and the considerably reduced reactivity provided by the Al coating.

Identification of Reaction Products:

To identify the compounds formed in the fiber-matrix reaction zone, X-ray diffraction analyses of the as-fabricated and thermally exposed DWA and TRW composites were performed. DWA composite samples were of SiC fiber reinforced Ti (A55) and also composites of Al, Mo and V coated Ti (A55). The TRW composites were of SiC fiber reinforced Ti (A55), Ti (3Al-2.5V) and Ti (6Al-4V). The experimental apparatus consisted of a Siemens diffractometer with a high intensity, fine focus, copper tube. The specimens were

rotated about the ϕ axis to reduce fluctuations in intensity which may result, due to the presence of large grains and texture within the sample. The data were collected using a diffracted beam graphite monochromator. The monochromator improved the peak-to-background ratios by minimizing contributions to the background from continuous radiation as well as fluorescence from the Ti-base substrate. Survey scans of 2θ were made at 1° per minute and slow scans were made at $1/4^\circ$ per minute.

The samples were ground using 600 grit paper to uniformly expose the filaments. They were then fine-polished on a lapping wheel using 1 micron and 0.3 micron alumina powder. In order to get a meaningful comparison in a semiquantitative analysis, the samples were ground to expose about 33% area fraction of filament in all cases. This procedure maximized the effective volume of the reaction zone so that small amounts of transformation about the interface could be observed with X-ray diffraction.

DWA composite samples in the as-received condition as well as those exposed to 1600°F and 1800°F for 25 hours each were examined to determine the various phases at each temperature. The results are presented in Tables 1 through 12. The 2θ values denoted by an asterisk represent unidentified reflections which were of low intensity. The intensity for each reflection was determined by integrating the area under the corresponding peak, using a small grid technique.

The as-fabricated composite gave diffraction lines from α -Ti, β -SiC and TiC. Although a few weak reflections appeared due to Ti_5Si_3 , the presence of Ti_5Si_3 phase, in general, in the

Table 1. - Experimental d-spacings and intensities for phases identified in as-fabricated SiC-Ti(A55)

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.02	2.6330	300	(???)				
35.61	2.5190	9,000		(111)	(010)	(111)	(002)
38.31	2.3474	615			(002)		
40.11	2.2461	830			(011)		
41.06	2.1963	75					(211)
41.46	2.1761	75				(200)	
41.96	2.1513	50		(200)			(300)
52.86	1.7305	325			(012)		
*59.01	1.5639	75					
60.01	1.5403	970		(220)			
60.76	1.5230	124				(220)	
62.86	1.4771	164			(110)		
70.46	1.3352	600			(103)		
71.81	1.3134	856		(311)			
72.61	1.3009	81				(113)	
75.46	1.2587	338		(222)		(222)	
75.98	1.2514	313			(112)		
77.21	1.2344	100			(201)		
82.16	1.1722	36			(004)		
86.56	1.1235	23			(202)		
92.46	1.0666	75			(014)		
100.06	1.0050	20				(331)	
100.81	.99958	162		(331)			
102.21	.98966	150			(203)		
*104.11	.97672	88					
104.71	.97276	81		(420)			
108.96	.94635	75			(211)		
114.01	.91836	188			(114)		
119.61	.89116	198		(422)	(212)		
121.06	.88472	119				(224)	

Table 1. - Continued

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
121.86	.88126	244			(015)		
126.11	.86404	32			(204)		
133.46	.83845	725		(333, 511)			
138.96	.82243	150			213)		

Table 2. - Experimental d-spacings and intensities for phases identified in 1600°F-25 hour exposed SiC-Ti(A55)

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.06	2.6300	150	(???)				
35.61	2.5190	8,500		(111)	(010)	(111)	(002)
36.91	2.4331	87					(210)
38.18	2.3551	650			(002)		
40.11	2.2461	910			(011)		
40.96	2.2015	164					(211)
41.86	2.1562	145		(200)		(200)	(300)
42.66	2.1190	32					(112)
52.81	1.7320	350			(012)		
60.06	1.5391	660		(220)			
60.66	1.5253	127				(220)	
61.41	1.5084	75					(222)
62.86	1.4771	262			(110)		
70.26	1.3386	600			(103)		
71.81	1.3134	574		(311)			
72.56	1.3017	101				(113)	
75.36	1.2601	191		(222)		(222)	
75.86	1.2531	210			(112)		
77.16	1.2351	117			(201)		
86.62	1.1229	25			(202)		
92.31	1.0680	71			(014)		
100.46	1.0023	68				(331)	
101.42	.99522	50		(331)			
102.01	.99105	252			(203)		
104.32	.97534	86		(420)			
106.00	.96445	20	(???)				
108.81	.94723	75			(211)		
113.56	.92072	125			(114)		
119.61	.89116	172		(422)	(212)		
120.61	.88669	108				(224)	

Table 2. - Continued

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
121.46	.88298	175			(015)		
*122.51	.87850	20					
133.41	.83861	318		(333, 511)			
138.86	.82270	100			(213)		

Table 3. - Experimental d-spacings and intensities for phases identified in 1800°F - 25 hour exposed SiC-Ti(A55)

2θ	d	Intensity (arbitrary units)	α-SiC	β-SiC	α-Ti	TiC	Ti ₅ Si ₃
34.16	2.6225	160	(???)				
35.61	2.5190	11,250		(111)	(010)	(111)	(002)
36.91	2.4331	163					(210)
37.53	2.3944	79					(102)
38.41	2.3415	810			(002)		
40.11	2.2461	1,557			(011)		
40.93	2.2030	290					(211)
41.93	2.1527	575		(200)		(200)	(300)
42.63	2.1190	215					(112)
53.03	1.7253	188			(012)		
*59.21	1.5591	50					
60.21	1.5356	470		(220)			
60.78	1.5230	302				(220)	
61.41	1.4656	75					(222)
63.01	1.4953	250			(110)		
65.56	1.4227	32					(321)
66.51	1.4046	100					(410) (213)
68.61	1.3666	50					(402)
70.71	1.3312	275			(103)		
71.83	1.3131	450		(311)			
72.76	1.2986	150				(113)	
75.46	1.2587	288		(222)		(222)	
76.26	1.2475	340			(112)		
77.36	1.2325	127			(201)		
82.36	1.1698	20			(004)		
83.86	1.1527	20					(502)
86.81	1.1209	36			(202)		
92.86	1.0631	52			(014)		

Table 3. - Continued

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
100.22	1.0039	40				(331)	
100.66	1.0007	89		(331)			
102.41	.98827	275			(203)		
104.36	.97507	200		(420)			
106.26	.96281	138	(???)				
109.06	.94576	125			(211)		
114.46	.91604	112			(114)		
119.86	.89003	156		(422)	(212)		
121.06	.88472	100				(224)	
*122.41	.87893	200					
133.36	.83876	325		(333, 511)			
139.56	.82083	110			(213)		

Table 4. - Experimental d-spacings and intensities for phases identified in as-fabricated SiC-Ti(A55)/Al

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.01	2.6337	250	(???)				
35.61	2.5190	11,400		(111)	(010)	(111)	(002)
38.08	2.3611	860			(002)		
40.05	2.2493	2,200			(011)		
40.82	2.2087	50					(211)
41.42	2.1781	100				(200)	
41.86	2.1562	25		(200)			(300)
52.81	1.7320	219			(012)		
*59.21	1.5591	50					
60.06	1.5391	813		(220)			
60.56	1.5276	253				(220)	
62.81	1.4872	210			(110)		
70.41	1.3361	550			(103)		
71.76	1.3142	958		(311)			
72.31	1.3055	171				(113)	
75.68	1.2556	601		(222)		(222)	
76.26	1.2475	108			(112)		
77.16	1.2351	155			(201)		
86.56	1.1235	41			(202)		
92.41	1.0671	47			(014)		
99.61	1.0083	20				(331)	
101.06	.99778	270		(331)			
102.01	.99105	310			(203)		
104.48	.97427	427		(420)			
105.81	.96566	71	(???)				
108.86	.94694	121			(211)		
113.98	.91862	143			(114)		
119.66	.89093	226		(422)	(212)		
120.61	.88669	238				(224)	

Table 4 . - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
121.81	.88148	246			(015)		
133.47	.83842	608		(333, 511)			
138.81	.82283	156			(213)		

Table 5. - Experimental d-spacings and intensities for phases identified in 1600°F-25 hour exposed SiC-Ti(A55)/Al

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.16	2.6225	150	(???)				
35.58	2.5210	10,950		(111)	(010)	(111)	(002)
36.76	2.4428	172					(210)
38.14	2.3575	972			(002)		
40.00	2.2521	3,435			(011)		
40.88	2.2056	178					(211)
41.88	2.1552	216		(200)		(200)	(300)
42.56	2.1223	82					(112)
52.72	1.7348	470			(012)		
*59.31	1.5567	70					
60.06	1.5391	584		(220)			
60.61	1.5265	187				(220)	
62.85	1.4773	277			(110)		
70.26	1.3386	455			(103)		
71.81	1.3135	545		(311)			
72.41	1.3040	120				(113)	
75.46	1.2587	308		(222)		(222)	
75.91	1.2523	310			(112)		
77.21	1.2345	108			(201)		
86.48	1.1244	71			(202)		
92.21	1.0689	110			(014)		
100.06	1.0050	25				(331)	
101.02	.99807	110		(331)			
101.91	.99175	144			(203)		
104.44	.97454	250		(420)			
106.06	.96408	94	(???)				
108.86	.94694	116			(211)		
113.64	.92029	209			(114)		
119.61	.89116	152		(422)	(212)		

Table 5. - Continued

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
120.68	.88638	128				(224)	
121.51	.88277	134			(015)		
122.26	.87956	50					
125.66	.86578	40			(204)		
133.38	.83870	422		(333, 511)			
136.51	.82925	50				(115)	
138.71	.82311	200			(213)		

Table 6 . - Experimental d-spacings and intensities for phases identified in 1800°F-25 hour exposed SiC-Ti(A55)/Al

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
33.92	2.6481	302	(???)				
35.72	2.5115	21,750		(111)	(010)	(111)	(002)
36.78	2.4415	414					(210)
37.58	2.3913	226					(102)
38.48	2.3374	230			(002)		
40.18	2.2424	1,800			(011)		
40.92	2.2035	298					(211)
41.98	2.1503	512		(200)		(200)	(300)
42.68	2.1166	190					(112)
53.06	1.7244	322			(012)		
*58.72	1.5710	165					
60.12	1.5377	1,050		(220)			
60.72	1.5239	680				(220)	
61.32	1.5105	120					(222)
*61.72	1.5016	124					
63.02	1.4737	622			(110)		
65.72	1.4196	24					(321)
66.52	1.4044	90					(410)
							(213)
68.68	1.3654	44					(402)
70.82	1.3293	417			(103)		
71.87	1.3124	1,425		(311)			
72.78	1.2983	120				(113)	
73.22	1.2916	145					(500)
							(004)
75.52	1.2578	670		(222)		(222)	
76.38	1.2458	245			(112)		
77.42	1.2317	129			(201)		

Table 6. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
83.92	1.1520	40					(502)
86.80	1.1210	40			(202)		
88.82	1.1007	20					(304)
90.20	1.0874	153				(400)	
*91.50	1.0753	153					
*93.02	1.0617	64					
99.32	1.0105	121				(331)	
101.32	.99593	767		(331)			
102.52	.98751	481			(203)		
104.52	.97401	136		(402)			
106.32	.96244	260	(???)				
109.12	.94541	467			(211)		
114.72	.91470	196			(114)		
119.02	.89385	230		(420)	(212)		
120.32	.88798	742				(224)	
*122.42	.87889	336					
125.42	.86672	44			(204)		
133.52	.83826	1,528		(333, 511)			
136.72	.82865	144				(115)	

Table 7. - Experimental d-spacings and intensities for phases identified in as-fabricated SiC-Ti(A55)/Mo

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.06	2.6300	130	(???)				
35.61	2.5190	12,480		(111)	(010)	(111)	(002)
36.86	2.4364	120					(210)
38.31	2.3474	2,210			(002)		
*38.81	2.3183	150					
40.13	2.2450	3,750			(011)		
40.96	2.2015	50					(211)
41.36	2.1811	25				(200)	
41.86	2.1562	25		(200)			(300)
52.86	1.7305	611			(012)		
60.01	1.5402	295		(220)			
60.56	1.5276	125				(220)	
62.86	1.4771	182			(110)		
70.48	1.3349	778			(103)		
71.86	1.3126	454		(311)			
72.46	1.3032	116				(113)	
75.48	1.2584	177		(222)		(222)	
76.01	1.2509	471			(112)		
77.28	1.2335	143			(201)		
81.98	1.1743	113			(004)		
86.56	1.1235	43			(202)		
92.46	1.0666	75			(014)		
100.16	1.0043	25				(331)	
100.86	.99922	92		(331)			
102.18	.98986	269			(203)		
104.43	.97460	154		(420)			
105.16	.96984	46	(???)				
108.96	.94634	116			(211)		
114.01	.91836	278			(114)		
119.36	.89230	132		(422)	(212)		

Table 7. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
120.41	.88758	114				(224)	
121.81	.88148	488			(015)		
133.46	.83845	373		(333, 511)			
139.16	.82190	189			(213)		

Table 8. - Experimental d-spacings and intensities for phases identified in 1600°F-25 hour exposed SiC-Ti(A55)/Mo

2θ	d	Intensity (arbitrary units)	α-SiC	β-SiC	α-Ti	TiC	Ti ₅ Si ₃
33.92	2.6405	400	(???)				
35.61	2.5190	13,140		(111)	(010)	(111)	(002)
36.76	2.4428	416					(210)
38.41	2.3415	647			(002)		
40.16	2.2435	930			(011)		
40.91	2.2040	216					(211)
41.91	2.1537	226		(200)		(200)	(300)
42.61	2.1199	53					(112)
53.01	1.7259	225			(012)		
*59.49	1.5529	50					
60.13	1.5374	2,812		(200)			
60.56	1.5276	750				(220)	
*62.01	1.4953	120					
62.96	1.4750	372			(110)		
70.76	1.3303	540			(103)		
71.91	1.3118	1,425		(311)			
72.51	1.3024	376				(113)	
75.56	1.2573	562		(222)		(222)	
76.11	1.2495	386			(112)		
77.36	1.2325	177			(201)		
82.11	1.1727	60			(004)		
86.81	1.1209	47			(202)		
90.12	1.0882	84				(400)	
*91.21	1.0780	110					
92.96	1.0622	88			(014)		
99.36	1.0102	50				(331)	
101.16	.99707	633		(331)			
102.41	.98827	420			(203)		
104.51	.97407	672		(420)			

Table 8. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
105.56	.96726	238	(???)				
108.91	.94664	184			(211)		
114.28	.91696	119			(114)		
119.06	.89367	100		(422)	(212)		
120.31	.88802	582				(224)	
121.41	.88319	249			(015)		
*122.26	.87956	196					
133.51	.83829	771		(333, 511)			
139.56	.82083	194			(213)		

Table 9 . - Experimental d-spacings and intensities for phases identified in 1800°F-25 hour exposed SiC-Ti(A55)/Mo

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.16	2.6225	375	(???)				
35.61	2.5190	11,430		(111)	(010)	(111)	(002)
36.81	2.4395	418					(210)
37.46	2.3987	167					(102)
38.41	2.3415	786			(002)		
40.16	2.2435	1,605			(011)		
40.93	2.2030	255					(211)
41.93	2.1527	432		(200)		(200)	(300)
42.61	2.1199	398					(112)
53.06	1.7244	226			(012)		
60.16	1.5368	814		(220)			
60.78	1.5230	528				(220)	
61.46	1.5074	161					(222)
63.01	1.4739	236			(110)		
65.71	1.4198	106					(321)
66.48	1.4052	200					(410)
							(213)
68.66	1.3658	84					(402)
70.81	1.3295	350			(103)		
71.91	1.3118	882		(311)			
72.78	1.2983	326				(113)	
73.21	1.2917	75					(500)
							(004)
75.63	1.2562	430		(222)		(222)	
76.46	1.2447	405			(112)		
77.46	1.2311	214			(201)		
78.46	1.2179	78					(420)
79.46	1.2051	70					(331)
83.86	1.1527	121					(502)
85.12	1.1388	30					(214)

Table 9. - Continued

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
86.16	1.1277	50					(511)
87.06	1.1183	125			(202)		
88.36	1.1052	55					(304)
*91.41	1.0761	73					
100.26	1.0036	50				(331)	
101.06	.99778	360		(331)			
102.46	.98792	372			(203)		
104.41	.97674	390		(420)			
106.31	.96250	378	(???)				
108.56	.94872	130			(211)		
*109.46	.94342	118					
114.51	.91578	127			(114)		
119.46	.89184	256		(422)	(212)		
120.71	.88625	434				(224)	
*122.51	.87850	464					
125.41	.86675	88			(204)		
133.51	.83829	394		(333, 511)			
136.61	.82807	140				(115)	

Table 10. - Experimental d-spacings and intensities for phases identified in as-fabricated SiC-Ti(A55)/V

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.06	2.6300	250	(???)				
35.61	2.5190	10,530		(111)	(010)	(111)	(002)
38.11	2.3593	1,350			(002)		
40.01	2.2515	1,521			(011)		
41.06	2.1963	120					(211)
41.56	2.1711	120				(200)	
42.02	2.1483	25		(200)			(300)
52.76	1.7835	307			(012)		
*59.31	1.5567	50					
60.01	1.5403	1,164		(220)			
61.01	1.5174	316				(220)	
62.81	1.4782	283			(110)		
70.51	1.3344	770			(103)		
71.76	1.3142	1,295		(311)			
72.56	1.3017	122				(113)	
75.46	1.2587	423		(222)		(222)	
76.16	1.2489	168			(112)		
*76.91	1.2385	40					
77.56	1.2298	50			(201)		
81.86	1.1757	102			(004)		
90.06	1.0887	50				(400)	
*91.46	1.0757	50					
*93.56	1.0570	50					
99.91	1.0061	20				(331)	
100.86	.99922	160		(331)			
102.01	.99105	131			(203)		
*104.11	.97672	104					
104.71	.97276	106		(420)			
105.91	.96502	75	(???)				

Table 10. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
109.06	.94576	20			(211)		
114.06	.91810	131			(114)		
119.11	.89343	125		(422)	(212)		
120.51	.88713	389				(224)	
121.91	.88104	210			(015)		
133.41	.83861	668		(333, 511)			

Table 11. - Experimental d-spacings and intensities for
phases identified in 1600°F-25 hour exposed SiC-Ti(A55)/V

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.06	2.6300	575	(???)				
35.66	2.5156	19,350		(111)	(010)	(111)	(002)
36.81	2.4395	300					(210)
38.51	2.3357	525			(002)		
40.26	2.2381	1,587			(011)		
40.96	2.2015	75					(211)
41.41	2.1786	120				(200)	
41.96	2.1513	150		(200)			(300)
42.62	2.1195	100					(112)
53.16	1.7214	230			(012)		
*59.11	1.5615	250					
60.16	1.5368	1,350		(220)			
60.71	1.5241	597				(220)	
*61.91	1.4975	120					
63.16	1.4708	386			(110)		
71.06	1.3238	918			(103)		
71.88	1.3123	1,350		(311)			
72.66	1.3001	210				(113)	
75.56	1.2573	615		(222)		(222)	
76.36	1.2461	433			(112)		
77.66	1.2284	175			(201)		
87.11	1.1178	100			(202)		
90.32	1.0863	70				(400)	
*91.50	1.0753	70					
92.96	1.0622	70			(014)		
99.96	1.0058	120				(331)	
101.01	.99814	696		(331)			
102.31	.98901	604			(203)		
104.61	.97342	786		(420)			

Table 11. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
106.26	.96281	186	(???)				
109.12	.94541	125			(211)		
*109.86	.94111	125					
114.61	.91526	109			(114)		
119.16	.89321	216		(422)	(212)		
120.16	.88869	777				(224)	
*122.31	.87935	524					
131.46	.84492	120			(300)		
133.51	.83829	1,411		(333, 511)			
136.46	.82940	142				(115)	
140.51	.81836	176			(213)		

Table 12. - Experimental d-spacings and intensities for phases identified in 1800°F-25 hour exposed SiC-Ti(A55)/V

2 θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
34.02	2.6330	350	(???)				
35.71	2.5121	11,250		(111)	(010)	(111)	(002)
36.91	2.4331	320					(210)
37.58	2.3913	184					(102)
38.51	2.3357	525			(002)		
40.26	2.2381	1,473			(011)		
40.98	2.2004	300					(211)
41.98	2.1503	350		(200)		(200)	(300)
42.66	2.1176	318					(112)
53.16	1.7214	303			(012)		
*54.36	1.6862	84					
*59.31	1.5567	125					
60.13	1.5374	625		(220)			
60.78	1.5230	552				(220)	
61.41	1.5084	222					(222)
63.16	1.4708	380			(110)		
65.71	1.4198	77					(321)
66.56	1.4037	423					(410)
							(213)
68.76	1.3640	75					(402)
71.06	1.3238	386			(103)		
71.91	1.3118	1,113		(311)			
72.71	1.2993	335				(113)	
73.50	1.2873	60					(500)
							(004)
75.56	1.2573	506		(222)		(222)	
76.51	1.2440	456			(112)		
77.61	1.2291	155			(201)		
78.56	1.2166	32					(420)

Table 12. - Continued

2θ	d	Intensity (arbitrary units)	α -SiC	β -SiC	α -Ti	TiC	Ti ₅ Si ₃
79.46	1.2051	45					(331)
82.51	1.1681	61			(004)		
83.91	1.1521	90					(502)
85.11	1.1389	35					(214)
86.06	1.1288	39					(511)
87.06	1.1183	110			(202)		
88.46	1.1042	81					(304)
*91.31	1.0770	52					
100.02	1.0053	60				(331)	
100.71	1.0003	294		(331)			
102.36	.99862	522			(203)		
104.56	.97375	433		(420)			
106.26	.96281	312	(???)				
*108.22	.95075	55					
109.01	.94605	39			(211)		
*109.66	.94226	50					
114.81	.91424	157			(114)		
119.96	.88958	198		(422)	(212)		
120.56	.88691	496				(224)	
*122.51	.87850	552					
*123.16	.87580	125					
125.56	.86617	135			(204)		
133.41	.83861	1,008		(333, 511)			
136.81	.82839	292				(115)	
140.56	.81824	188			(213)		

as-fabricated specimens is doubtful. The Ti_5Si_3 phase started appearing more clearly in all samples after an exposure of 1600°F for 25 hours. With further thermal exposure, the total intensity from Ti_5Si_3 was found to increase. A quantitative analysis of the phases found in DWA samples was not attempted at the present time.

Some moderate line shifts were observed in the α -Ti diffraction patterns after the thermal exposure. This would be expected due to lattice expansion or contraction as C and/or Si goes into solution. Residual stresses in the composite, introduced during thermal exposure, may have also contributed to the α -Ti shifts.

No lines from the interfacial coatings could be detected in the diffraction patterns of the coated composites. This was probably because the various coatings went into solution with the matrix during composite fabrication cycle.

Composites of SiC reinforced Ti (A55), Ti (3Al-2.5V) and Ti (6Al-4V) fabricated by TRW were found to be well bonded compared to those of DWA. These composites were thermally exposed to 1200°F , 1600°F and 1800°F for 25 hours each. The most obvious difference in the TRW composites of Ti (A55) compared to DWA composites of Ti (A55) was that lines due to TiSi and TiSi_2 were present in the TRW composites for 1600°F and 1800°F exposures, whereas, in the DWA composite, these lines were absent. In addition, reflections due to β -Ti were identified in the as-fabricated and thermally exposed composites of Ti (3Al-2.5V) and Ti (6Al-4V).

Quantitative Study of Reaction Products:

For the TRW composites, the study of reaction at the fiber-matrix interface is under study on a quantitative basis. One might expect that the interface reaction can be followed by studying the decrease in the integrated intensity of SiC reflections. However, with an increase in the thermal exposure, SiC reflections were found to overlap with the reflections of reaction products. In order to follow the intensity variation of SiC reflections, it is essential to separate the SiC reflections from other overlapping reflections. The Pearson VII function can be used in the separation of overlapping reflections, and this is given by

$$Y = Y_0 \left[1 + \frac{(X - X_0)^2}{ma^2} \right]^{-m}$$

where X_0 is the 2θ value corresponding to the peak height Y_0 , "a" is related to peak half-width and m is the shape factor. For $m = 1$, the variation is Cauchy, and for $m = \infty$, the variation is Gaussian.

A modified IBM nonlinear least squares program was obtained from Purdue University. This was originally written to separate two overlapping peaks. We have modified it so that it is now capable of separating 12 overlapping reflections from the $K\alpha_1$ - $K\alpha_2$ doublet. The program has been used for investigating the Ti (A55) composite samples of TRW. A typical curve is shown in Fig. 2 where individual reflections are separated using the program.

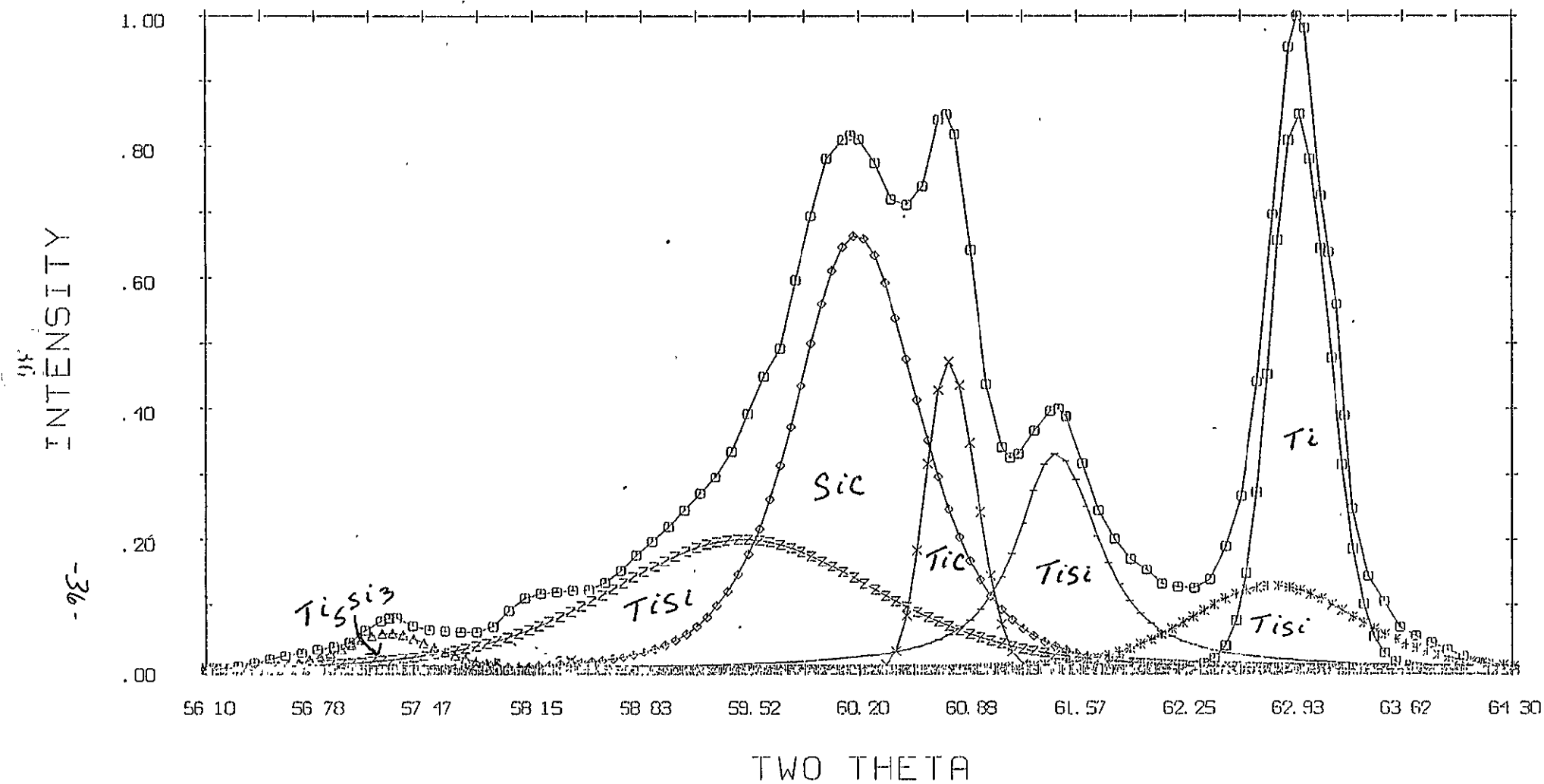


Fig. 2 SiC (220) reflection overlapped with reaction products in Ti (A55) composite exposed at 1600°F for 25 hours.

The half-widths and integrated intensity results are scattered when the same line is examined after the various thermal exposures. This effect may be due to the development of a preferred orientation. More accurate data are being collected at this time. This is required before any conclusions can be made at this time.

Publications and Talks:

The research on this study resulted in the X-ray data listed in Tables 1 through 12, and the following Master's Thesis: A Study of Interfacial Reactions in Silicon Carbide Reinforced Titanium, George Washington University, 1979. These results will be included in future publications.

PART 3 - SOLUTIONS FOR DIFFUSION IN SINGLE-, TWO-,
AND THREE-PHASE BINARY ALLOY SYSTEMS

OBJECTIVE

In the analysis of diffusion controlled processes (e.g., the diffusion of planar protective coatings into substrate materials, the filament-matrix interaction in metal matrix composites, or the homogenization of powder compacts), the ability to predict the degree of interaction between different components, given some exposure conditions, is needed. Solutions of the diffusion equation have been reported for several different initial and boundary conditions. Most of these, however, are restricted to one geometry, applicable only for infinite or semi-infinite systems, or require that the diffusion coefficient be independent of concentration. The objective of this study is to develop general solutions for treating diffusion in single-, two-, and three-phase binary alloy systems.

APPROACH

Finite-difference solutions were developed in this study for treating one-dimensional transient diffusion in single-, two-, and three-phase binary alloy systems. These solutions are applicable for planar, cylindrical, or spherical geometries with any diffusion-zone size and any continuous (within each phase) variation of the diffusion coefficient with concentration. Special techniques were included to account for differences in molal volumes, initiation and growth of an intermediate phase (three-phase system, disappearance of a phase (two-, and three-phase systems), and the presence of

an initial composition profile in the specimen. In each analysis, an effort was made to achieve good accuracy while minimizing computation time.

RESULTS AND DISCUSSION

Diffusion calculations were performed (ref. 3) to establish the conditions under which concentration dependence of the diffusion coefficient was important in single-, two-, and three-phase binary alloy systems. Finite-difference solutions for each type of system were obtained using diffusion coefficient variations typical of those observed in real alloy systems. Solutions were also obtained using average diffusion coefficients determined by taking a logarithmic average of each diffusion coefficient variation considered. The solutions for constant diffusion coefficients were used as references in assessing the effects of diffusion coefficient variations. Calculations were performed for planar, cylindrical, and spherical geometries in order to compare the effect of diffusion coefficient variations with the effect of interface geometries.

Diffusion coefficient variations in single-phase systems and in the major-alloy phase of two-phase systems were found to effect the kinetics of diffusion as strongly as the interfacial geometry of the diffusion couple. Concentration dependence of the diffusion coefficient in the minor-alloy phase of a two-phase system was found to have only an initial transient effect on the diffusion kinetics. In three-phase systems, the intermediate phase did not increase in thickness if the diffusion coefficient of the inter-

mediate phase was smaller than the diffusion coefficient of the major-alloy phase. Under these conditions, the three-phase diffusion problem could be treated as a two-phase problem. However, if the diffusion coefficient of the intermediate phase was larger than the diffusion coefficient of the major-alloy phase, the intermediate phase grew rapidly at the expense of the major and minor phases. In most of the cases considered, the diffusion coefficient of the major-alloy phase was the key parameter that controlled the kinetics of interdiffusion.

A semiempirical relationship was developed (refs. 4,5) which describes the extent of interaction between constituents in single-phase binary alloy systems having planar, cylindrical, or spherical interfaces. This relationship makes possible a quick estimate of the extent of interaction without lengthy numerical calculations. It includes two parameters which are functions of mean concentration and interface geometry. Experimental data for the copper-nickel system are included to demonstrate the usefulness of this relationship.

PUBLICATIONS AND TALKS

The research on this study resulted in three publications: Effect of Concentration Dependence of the Diffusion Coefficient on Homogenization Kinetics in Multiphase Binary Alloy Systems, NASA TP 1281, 1978; Geometric Relationships for Homogenization in Single-Phase Binary Alloy Systems, NASA TP 1349, 1978; and An Empirical Relationship for Homogenization in Single-Phase Binary Alloy Systems, Met. Trans., Vol. 10A, No. 3, 1979.

PART 4 - MOISTURE EFFECTS ON GRAPHITE/POLYIMIDE COMPOSITES

OBJECTIVE

Composite materials consisting of graphite fibers in a polyimide resin matrix offer potential for considerable weight savings in structural applications requiring high temperature service. The space shuttle orbiter vehicle aft body flap is a prime candidate for possible application, where operating temperatures will be too high for graphite epoxy consideration and considerable weight savings may be possible compared to conventional aluminum structure. However, the characterization of graphite polyimide materials with respect to mechanical property degradation by environmental effects has not yet been adequately accomplished. The objective of this study was to do such a characterization.

APPROACH

The Materials Research Branch undertook this problem in support of NASA's CASTS program. The experimental program included a large matrix of both material and environmental variables.

<u>Material</u>	HTS2/PMR 15, Celion/PMR 15
<u>Orientation</u>	(0,+45,90) ₂ , (0) ₈
<u>Moisture Condition</u>	As processed, vacuum dried, saturated, saturated and thermally cycled
<u>Temperature</u>	117, 294, 589K (-250, 70, 600°F)
<u>Property</u>	Tension, compression, interlaminar shear, rail shear, flexure

The grant's main contribution to this program was to develop software for a Hewlett-Packard Data Acquisition System to permit extensive

mechanical property data collection on the graphite/polyimide composites.

RESULTS AND DISCUSSION

Software Development:

Three computer programs were developed for an HP-9845 system to facilitate real-time data collection. A listing of these programs is given in Appendix A. Program 1 utilizes a digital voltmeter for data collection during static tests. Program 2 utilizes a fast response system voltmeter and is used for fatigue tests. Two amplifiers are used with the system voltmeter to improve accuracy. Programs 1 and 2 store the data on magnetic tape cartridge. This data can subsequently be read using program 3 which gives stress-strain and modulus-strain plots, as required. The programs were written to accommodate more than one test machine at the Langley Research Center. Either a Tinius-Olsen testing machine or an MTS machine could be used with the data acquisition system. All the programs are documented to facilitate usage by other researchers.

Mechanical Properties:

The general conclusions resulting from the on-going study on graphite/polyimide composites are as follows (ref. 6): Moisture conditioning produced moderate to severe reduction in compressive and interlaminar shear properties at 589 K (600°F). No reduction was observed in tests at room temperature or at 117 K (-250°F). The compressive property appeared to be affected more than the interlaminar shear property. Vacuum drying, on the other hand, appeared to improve both properties at elevated temperatures and

might be considered for final processing if it can be shown that increases would not be lost by environmental degradation in service. The degradation by moisture conditioning appears to be associated with the lowering of the glass transition temperature, and this can occur after only a few weeks exposure to condensing humidity conditions at 355 K (180°F).

PUBLICATIONS AND TALKS

The research on this study resulted in the three computer programs listed in Appendix A, and the following technical paper: Mechanical Property Degradation of Graphite/Polyimide Composites after Exposure to Moisture or Shuttle Orbiter Fluids, NASA CP 2079, 1979.

SUMMARY

Good progress has been made in several areas. Mechanical property degradation and chemical interactions were studied in titanium matrix composites. The results suggest that an interfacial coating of aluminum considerably reduces the reaction between SiC fiber and titanium matrix. General finite difference solutions were developed to treat diffusion in multiphase binary alloy systems with planar, cylindrical or spherical interface. Software was developed for the Hewlett-Packard data acquisition system to facilitate real-time stress-strain data collection on graphite/polyimide composites. The research on this grant resulted in six publications.

REFERENCES

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2. L. J. House: A Study of Interfacial Reactions in Silicon Carbide Reinforced Titanium, M.S. Thesis, George Washington University, 1979.
3. D. R. Tenney and J. Unnam: Effect of Concentration Dependence of the Diffusion Coefficient on Homogenization Kinetics in Multiphase Binary Alloy Systems, NASA TP 1281, 1978.
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5. J. Unnam, D. R. Tenney, and B. A. Stein: An Empirical Relationship for Homogenization in Single-Phase Binary Alloy Systems, Met. Trans., Vol. 10A, No. 3, 1979.
6. W. B. Lisagor: Mechanical Property Degradation of Graphite/Polyimide Composites after Exposure to Moisture or Shuttle Orbiter Fluids, NASA CP 2079, 1979.

APPENDIX A

HP-9845 Computer Programs

Program 1.

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10 ! Program#1:Static Test with Tinius-Olsen or MTS machine.
20 OPTION BASE 1
30 DEG
40 DIM C(20),C(100,20),A$(50),I(20),H(20),P(20),F$(5),R(20)
50 COM R(0:8)
60 INPUT "1 For CPT output, 2 For PAPER output: Select.",Prt
70 IF Prt=1 THEN PRINTER IS 16
80 IF Prt=2 THEN PPINTER IS 0
90 C(1)=1
100 C(14)=50000
110 R(7)=1
120 R(8)=3
130 INPUT "Total # of Channels(1,3,.....20)?",R(1)
140 IF (R(1)<2) OR (R(1)>20) THEN GOTO 130
150 R(2)=INT(2000/R(1))
160 REDIM G(R(2),R(1))
170 INPUT "# of Strain Gages(0,1,...18)?",C(12)
180 INPUT "# of Extensometers(0,1,2)?",C(11)
190 IF C(11)+C(12)+2>R(1) THEN GOTO 170
200 INPUT "Type of Test, Sp.#, Date, etc. (50cha.)?",A$
210 PPINT LIN(2)
220 PRINT "Program 1 Output."
230 PPINT LIN(1)
240 PRINT A$
250 PRINT LIN(1)
260 FIXED 0
270 PRINT "-----"
280 PPINT "channels =";R(1)
290 PRINT "points/cha. =";R(2)
300 GOSUB Init
310 C(5)=R(1)
320 C(7)=R(2)
330 R(4)=INT((R(1)-2)/2)
340 IF R(4)<1 THEN R(4)=1
350 INPUT "1 for Tinius-Olsen, 2 for MTS : Select.",C(13)
360 IF C(13)=2 THEN INPUT "How many lbs does the full scale (10V) correspond
to? (5K,10K,20K,25K,50K,100K)",C(14)
370 INPUT "What is the maximum load (lbs) you expect to use?",R(0)
380 R(8)=10*(C(14)-R(0))
390 IF R(8)<1 THEN R(8)=2
400 IF R(8)<.1 THEN R(8)=1
410 R(9)=20
420 R(R(4)+1)=R(8)
430 H(R(4)+1)=R(3)
440 P(R(4)+1)=1
450 FOR I=1 TO R(4)
460 R(I)=R(7)
470 H(I)=R(8)+I
480 P(I)=I+1
490 NEXT I
500 IF R(1)=2 THEN GOTO 570
510 FOR I=R(4)+2 TO R(1)
520 R(I)=R(7)
530 H(I)=R(8)+I-1
540 P(I)=I
550 NEXT I
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570   Ru(0)=-1
580   Ru(5)=Ru(7)
590   INPUT "Channel # to be Balanced: 20, 21, 22, ..., 39?", Ru(0)
610   IF Ru(0)=-1 THEN GOTO 570
620   IF Ru(0)=20 THEN Ru(5)=Ru(8)
630   OUTPUT 722 USING 1390;INT(Ru(5))
640   OUTPUT 709 USING "#,K";"C"
650   OUTPUT 709 USING 1380;Ru(0)
660   GOTO 570
670   INPUT "Specimen Thickness in inches ?", C(2)
680   INPUT "Specimen Width in inches ?", C(3)
690   INPUT "Strain Gage Factor ?", C(4)
700   INPUT "Total Test Time (4 to 30 mins) ?", C(5)
710   INPUT "Tape Cartridge # ?", Ru(0)
720   INPUT "Specify 2-digit file# (10, 11, ..., 99)?", File
730   IF (File<10) OR (File>99) THEN GOTO 710
740   File=INT(File)
750   F$="FILE"&VAL$(File)
760   PRINT "Cartridge# =";Ru(0)
770   PRINT "Data File# =";File
780   PRINT "Strain Gages=";C(12)
790   PRINT "Extensometers=";C(11)
800   PRINT "-----"
810   PRINT LIN(1)
820   PRINT "To Stop: PAUSE , S=1 , EEXECUTE , CONTINUE."
840   S=0
850   Ru(6)=C(5)*60000-C(7)-C(8)+120
860   IF Ru(6)<0 THEN Ru(6)=0
870   OUTPUT 722;"T3"
880   INPUT "Press CONTINUE to Collect Data", Ru(0)
890   BEEP
900   FOR I=1 TO Ru(1)
910   OUTPUT 722 USING 1390;INT(Ru(I))
920   OUTPUT 709 USING "#,K";"C"
930   OUTPUT 709 USING 1380;N(I)
940   TRIGGER 722
950   ENTER 722 USING 1400;I(P(I))
960   NEXT I
970   FOR J=1 TO Ru(2)
980   IF S=1 THEN GOTO Stop
990   FOR I=1 TO Ru(1)
1000  OUTPUT 722 USING 1390;INT(R(I))
1010  OUTPUT 709 USING "#,K";"C"
1020  OUTPUT 709 USING 1380;N(I)
1030  TRIGGER 722
1040  ENTER 722 USING 1400;G(J,P(I))
1050  NEXT I
1060  WAIT Ru(6)
1070  NEXT J
1080 Stop: BEEP
1090  C(8)=J-1
1100  INPUT "Press CONTINUE to Store", Ru(0)
1110  C(9)=I(Ru(1))
1120  IF NOT ((C(9)<.0115) OR (C(9)>.0125)) THEN GOTO 1150
1130  FIXED 1
1140  PRINT "Bridge Volt.=";C(9)*500
1150  FIXED 0
1160  PRINT "# of Data Points Collected = ";C(8)
1170  FOR J=1 TO C(8)
1180  FOR I=1 TO Ru(1)-1
1190  G(J,I)=G(J,I)-I(I)
1200  NEXT I
1210  NEXT J
1220  INPUT "Insert Cartridge in RECOPD mode for data storage.", Ru(0)
1230  Ru(0)=C(6)+C(8)-70*-8+300*256
1240  CREATE F$,Ru(0)

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1250 REDIM G(C(5),C(8))
1255 ASSIGN #9 TO F#
1260 CHECK READ #9
1280 PRINT #9;C(*),G(*),A#
1290 ASSIGN #9 TO +
1300 Rv(0)=-1
1310 INPUT "More data collection(1 yes,0 no)?",Rv(0)
1320 IF Rv(0)<1 THEN GET "PROG3"
1330 GOTO 130
1340 END
1350 Init: 1
1360 REMOTE 7
1370 RESET 7
1380 IMAGE #,2Z,"E"
1390 IMAGE "R",1D
1400 IMAGE F
1410 OUTPUT 722;"FIT1"
1420 RETURN

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Program 2

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10 ! Program#2:Fatigue Test with Tinius-Olsen or MTS machine.
20 OPTION BASE 1
30 DEG
40 DIM C(20),G(100,2),A$(50),P(50),F$(6),Z$(50),Buffer$(1401)
50 COM Rv(0:21)
60 INPUT "1 for CRT output, 2 for PAPER output: Select.",Prt
70 IF Prt=1 THEN PRINTER IS 15
80 IF Prt=2 THEN PRINTER IS 0
90 FIXED 0
100 C(1)=2
110 Rv(1)=2
120 Rv(2)=100
130 Rv(3)=50
140 INPUT "Type of Test, Sp.#, Date, etc. (50cha.) ",A$
150 PRINT LINE(2)
160 PRINT "Program 2 Output."
170 PRINT A$
180 PRINT "-----"
190 PRINT "2 channels."
200 PRINT "100 points/chan."
210 PRINT "50 intervals."
220 INPUT "Cycles/second ?",C(10)
230 PRINT "Cycles/second=";C(10)
240 INPUT "GAIN for LOAD channel ?",Rv(10)
250 INPUT "GAIN for STRAIN channel ?",Rv(11)
260 PRINT "GAIN for LOAD channel=";Rv(10)
270 PRINT "GAIN for STRAIN channel=";Rv(11)
280 PRINT "-----"
290 C(12)=1
300 C(11)=0
310 INPUT "1 for Tinius-Olsen, 2 for MTS: Select.",C(13)
320 C(14)=50000
330 IF C(13)=2 THEN INPUT "To how many lbs does the full scale(10V) correspond to?(5K,10K,20K,25K,50K,100K)",C(14)
340 OUTPUT 9;"U2H,U2=12"
350 IMAGE "P",12
360 IMAGE #,22,"E"
370 INPUT "SVM Range (1 for 0.1V, 2 for 1V, 3 for 10V) ?",Rv(9)
380 REMOTE 7
390 RESET 7
400 OUTPUT 726 USING "#,K";"D.01S,N9999S,F1T1"
410 OUTPUT 726 USING 350;INT(Rv(9))
420 C(6)=Rv(1)
430 C(8)=C(7)=Rv(2)
440 C(9)=.012
450 P(1)=30
460 P(2)=60
470 P(3)=100
480 P(4)=1000
490 P(5)=10000
500 FOR I=6 TO Rv(3)
510 P(I)=10000+P(I-1)
520 NEXT I
530 Rv(0)=-1
540 INPUT "Channel # to be Balanced (0,1) ",Rv(0)
550 IF Rv(0)=-1 THEN GOTO 500
570 OUTPUT 709 USING 360;Pv(0)

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580 TRIGGER 726
590 GOTO 530
600 INPUT "Specimen Thickness in inches ?",C(2)
610 INPUT "Specimen Width in inches ?",C(3)
620 INPUT "Strain Gage Factor ?",C(4)
630 INPUT "Tape Cartridge# ?",Pv(19)
640 INPUT "Specify 2-Digit File# for Storing Data ('10,11,...80') ?",File
650 IF (File<10) OR (File>80) THEN GOTO 640
670 PRINT "To Store: PAUSE, S=1, EXECUTE, CONTINUE."
680 PRINT LIN(1)
690 PRINT "To Stop : PAUSE, S=2, EXECUTE, CONTINUE."
700 PRINT LIN(1)
710 S=0
720 OUTPUT 726;"T3"
730 FIXED 0
740 INPUT "Press CONTINUE to Collect Data",Rv(0)
750 OUTPUT 9;"U2G"
760 OUTPUT 9;"U2V"
770 ENTER 9;V
780 V=V+C(10)*1000
790 FOR K=1 TO Rv(3)
800 OUTPUT 709;"L01F0011"
820 OUTPUT 726 USING "#,K";"F1, T1,D.0010SH200S"
830 Buffer$=""
840 BEEP
850 ENTER 726 BFHS 1401;Buffer$
860 BEEP
870 OUTPUT 726;"T3"
880 Rv(18)=Rv(16)=-1E99
890 Rv(17)=Rv(15)=1E99
900 Buffer$(1400)=", "
910 J=15
920 FOR I=2 TO Rv(2)
930 G(I,1)=VAL(Buffer$(J))
940 G(I,1)=ABS(G(I,1)/Pv(10))
950 J=J+7
960 G(I,2)=VAL(Buffer$(J))
970 G(I,2)=ABS(G(I,2)/Pv(11))
980 J=J+7
990 IF Rv(18)<G(I,2) THEN Pv(18)=G(I,2)
1000 IF Rv(16)<G(I,1) THEN Rv(16)=G(I,1)
1010 IF Rv(17)>G(I,2) THEN Rv(17)=G(I,2)
1020 IF Rv(15)>G(I,1) THEN Rv(15)=G(I,1)
1030 NEXT I
1040 G(1,1)=G(2,1)
1050 G(1,2)=G(2,2)
1060 Rv(21)=Rv(18)-Rv(17)
1070 Rv(20)=Rv(16)-Rv(15)
1080 Rv(20)=Rv(20)/10*C(14)
1090 Pv(20)=Rv(20)/(C(2)+C(3))*1000
1100 Rv(21)=4+Rv(21)/(C(4)+500+C(9))
1110 Rv(20)=ABS(Rv(20)/Rv(21))
1120 PRINT "Cycle# =";INT(V)
1130 PRINT "Modulus=";Rv(20);"1/si."
1140 IF S=2 THEN GOTO Stop
1150 IF S=1 THEN GOSUB Store
1160 OUTPUT 9;"U2V"
1170 ENTER 9;V
1180 V=V+C(10)*1000
1190 IF V<Pv(K) THEN GOTO 1140
1200 NEXT K
1210 Stop:GET "PROG3"
1220 END
1230 Store:INPUT "Insert Cartridge in RECORD mode for data storage.",Pv(9)
1240 File=INT(File)
1250 F$="FILE"&VAL$(File)

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1260 Z$="Cartridge"&VAL$(RUC(19))&"; "&F$
1270 CREATE F$,10
1280 ASSIGN #9 TO F$
1295 CHECK READ #9
1300 PRINT #9;C(*),G(*),H$
1310 ASSIGN #9 TO +
1320 PRINT "-----"
1330 PRINT Z$
1340 PRINT "-----"
1350 File=File+1
1360 S=0
1370 RETURN

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Program 3

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10  ! Program#3: Retrieving & Plotting Data. May 11, 1979.
20  OPTION BASE 1
30  DEG
40  DIM C(20),F$(5)
50  COM Rv(0:37)
60  INPUT "1 for CRT plot, 2 for 9872 plot: Select.",Crt
70  INPUT "1 for CRT output, 2 for PAPER output: Select.",Prt
80  IF Crt=1 THEN PLOTTER IS 13,"GRAPHICS"
90  IF Crt=1 THEN GRAPHICS
100  Savep=0
110  IF Crt=1 THEN GOTO 140
120  OUTPUT 705;"IP 0.0,10500.3300"
130  PLOTTER IS 7,5,"9872A"
140  IF Prt=2 THEN PRINTER IS 0
150  IF Prt=1 THEN PRINTER IS 15
160  INPUT "Tape Cartridge #",Pv(0)
170  FIXED 0
180  Pread: INPUT "Specify 2-digit file# (10,11....,39).",File
190  IF (File<10) OR (File>99) THEN GOTO Pread
200  File=INT(File)
210  F$="FILE"&VAL$(File)
220  ASSIGN #9 TO F$
230  READ #9;C(*)
240  DIM G(40,50),X(0:39),Y(0:39),F(0:39),W(0:39),Wp(0:39)
250  DIM A$(50),X$(30),Y$(30),Z$(50)
260  Rv(1)=C(5)
270  Rv(2)=C(8)
280  REDIM G(Rv(2),0:Rv(1)-1)
290  FIXED 0
300  PRINT LIN(2)
310  Rv(0)=INT(Rv(0))
320  Z$="Cartridge"&VAL$(Rv(0))&": "&F$
330  PRINT "Program 3 Output: ",Z$
340  PRINT "# of channels=";Rv(1);"; points=cha.=";Pv(2)
350  PRINT LIN(1)
360  Rv(3)=0
370  Rv(33)=0
380  Rv(34)=0
390  Read: ASSIGN #9 TO F$
400  X$="STRAIN, in-in"
410  READ #9;C(*),G(*),A$
420  FIXED 5
430  C(9)=C(9)+500
440  IF Rv(33)=4 THEN GOTO Plot2
450  Y$="STRESS, ksi"
460  IF Pv(3)=1 THEN GOTO 790
470  Pv(20)=0
480  INPUT "Is Data to be Printed (1 yes, 0 no)?",Pv(20)
490  IF Rv(20)=0 THEN GOTO 590
500  FOR J=1 TO C(8)
510  FIXED 0
520  PRINT J;
530  FOR I=0 TO Rv(1)-1
540  FIXED 5
550  PRINT G(J,I);

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560 NEXT I
570 PRINT LIN(0)
580 NEXT J
590 IF Rv(3)=1 THEN GOTO 790
600 Rv(3)=1
610 FIXED 0
620 PRINT "-----"
630 PRINT "Load Channel = 0"
640 F(0)=1
650 IF C(12)=0 THEN GOTO 700
660 FOR I=1 TO C(12)
670 PRINT "Strain Gage =";I
680 F(I)=2
690 NEXT I
700 K=3
710 IF C(11)=0 THEN GOTO 770
720 FOR I=C(6)-C(11)-1 TO C(6)-2
730 PRINT "Extensometer =";I
740 F(I)=K
750 K=K+1
760 NEXT I
770 IF C(6)*C(11)+C(12)-1 THEN PRINT "Bridge Volt. =";C(6)-1
780 PRINT "-----"
790 FIXED 2
800 PRINT LIN(1)
810 PRINT "-----"
820 PRINT LIN(1)
830 PRINT "          Channel#    Wt. Fr."
840 PRINT LIN(1)
850 FOR I=0 TO Rv(1)-2
860 Y(I)=-1
870 INPUT "Chan.# & Weighting Factor for Y-axis",Y(I)
880 IF Y(I)=-1 THEN GOTO 940
890 INPUT "Wy(I) ?",Wy(I)
900 PRINT "Y-axis:      ";Y(I);"      ";Wy(I)
910 Rv(4)=I+1
920 NEXT I
930 PRINT LIN(1)
940 FOR I=0 TO Rv(1)-2
950 X(I)=-1
960 INPUT "Chan.# & Weighting Factor for X-axis",X(I)
970 IF X(I)=-1 THEN GOTO 1030
980 INPUT "Wx(I) ?",Wx(I)
990 PRINT "X-axis:      ";X(I);"      ";Wx(I)
1000 Rv(6)=I+1
1010 NEXT I
1020 PRINT LIN(1)
1030 INPUT "Are axes to be plotted(0,1)?",Rv(28)
1040 INPUT "pen # to use(1,4)?",Rv(11)
1050 PEN Rv(11)
1060 Plot2:FOR J=1 TO C(8)
1070 Rv(35)=0
1080 FOR I=0 TO Rv(4)-1
1090 Rv(8)=G(J,Y(I))
1100 Rv(21)=F(Y(I))
1110 GOSUB Function
1120 Rv(35)=Rv(35)+Rv(8)+Wy(I)
1130 NEXT I
1140 Rv(8)=G(J,Y(0))=Rv(35)/Rv(4)
1150 Rv(35)=0
1160 FOR I=0 TO Rv(6)-1
1170 Rv(8)=G(J,X(I))
1180 Rv(21)=F(X(I))
1190 GOSUB Function
1200 Rv(35)=Rv(35)+Rv(8)+Wx(I)
1210 NEXT I

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1220  Rv(8)=G(J,K(0))=Rv(35)
1230  NEXT J
1240  IF Rv(33)=4 THEN GOTO Modulus
1250  IF Rv(28)<1 THEN GOTO Plot
1260 Plot1:IF Crt=C THEN INPUT "Place 8.5x11 paper (horizontally) touching lower
-left-corner of plotter",Rv(0)
1270  FIXED 5
1280  GOSUB Maxmin
1290  PPINT "Xmin=";Rv(13)
1300  PRINT "Xmax=";Rv(12)
1310  PRINT "Ymin=";Rv(15)
1320  PRINT "Ymax=";Rv(14)
1330  INPUT "Xmin?",Rv(25)
1340  INPUT "Xmax?",Rv(24)
1350  INPUT "Ymin?",Rv(27)
1360  INPUT "Ymax?",Rv(26)
1370  INPUT "X-tic Interval?",Rv(9)
1380  INPUT "# of tics per X-unit label?",Rv(18)
1390  INPUT "X-axis label(30 Characters)?",X$
1400  INPUT "Y-tic Interval?",Rv(10)
1410  INPUT "# of tics per Y-unit label?",Rv(19)
1420  INPUT "Y-axis label(30 Characters)?",Y$
1430  Rv(15)=(Rv(24)-Rv(25))/6.5*10
1440  Rv(17)=(Rv(26)-Rv(27))/6.5*10
1450  IF Crt=2 THEN INPUT "P1,P2: 1 you set, 0 program selects, 2 old values re-
tained.",Savep
1460  IF Savep=0 THEN LOCATE 15,115,15,85
1470  IF Savep>1 THEN GOTO 1500
1480  DISP "Set P1, Press ENTER on plotter. Set P2, Press ENTER."
1490  LOCATE
1495  DISP ""
1500  SCALE Rv(25),Rv(24),Rv(27),Rv(26)
1510  AXES Rv(9),Rv(10),Rv(25),Rv(27),Rv(18),Rv(19),4
1520  Savep=2
1530  CSIZE 3,1/1.5
1540  LOG 5
1550  LDIR 0
1560  FIXED 3
1570  Xtics=INT((Rv(24)-Rv(25))/Rv(9))+.1
1580  Ytics=INT((Rv(26)-Rv(27))/Rv(10))+.1
1590  FOR I=0 TO Xtics STEP Rv(18)
1600  Xval=Rv(25)+I+Rv(9)
1610  MOVE Xval,Rv(27)-Rv(17)*3
1620  LABEL USING "k";Xval
1630  NEXT I
1640  MOVE (Rv(24)-Rv(25))/2,Rv(27)-7*Rv(17)
1650  LABEL USING "K";X$
1660  FIXED 0
1670  LOG 8
1680  FOR I=0 TO Ytics STEP Rv(19)
1690  Yval=Rv(27)+I+Rv(10)
1700  MOVE Rv(25)-Rv(15),Yval
1710  LABEL USING "k";Yval
1720  NEXT I
1730  LOG 5
1740  LDIR 90
1750  MOVE Rv(25)-9*Rv(18),(Rv(26)-Rv(27))/2
1760  LABEL USING "K";Y$
1770  PLOT Rv(25),Rv(26),-2
1780  PLOT Rv(24),Rv(25),-1
1790  PLOT Rv(24),Rv(27),-1
1800  PLOT Rv(25),Rv(26),-2
1810  LDIR 0
1820  MOVE (Rv(24)-Rv(25))/2,(Rv(27)-7*Rv(17)+Rv(26)
1830  LABEL USING "k";R$
1840  MOVE (Rv(24)-Rv(25))/2,(Rv(27)-7*Rv(17)+Rv(26)

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1850 LABEL USING "K";Z$
1860 Plot:PLOT G(1,X(0)),G(1,Y(0)),-2
1870 FOR J=1 TO C(8)
1880 Rv(0)=-1
1890 IF (G(J,Y(0))<1E-9) OR (G(J,X(0))<1E-9) THEN Rv(0)=-2
1900 PLOT G(J,X(0)),G(J,Y(0)),Rv(0)
1910 NEXT J
1920 IF Crt=1 THEN DUMP GRAPHICS
1930 IF Crt=1 THEN GCLEAP
1940 PEN 0
1950 IF Rv(34)>=1 THEN GOTO Read
1960 IF Rv(33)=0 THEN GOTO 2040
1970 IF NOT (Rv(33)=1) OR (Pv(33)=2) OR (Rv(33)=4) THEN GOTO 2000
1980 Rv(33)=0
1990 GOTO Read
2000 IF NOT (Rv(33)=3) THEN GOTO 2040
2010 Rv(31)=2
2020 Rv(33)=4
2030 GOTO Read
2040 IF Y(0)=0 THEN INPUT "Modulus(0No,1Tan,2Sec,3Tan&Sec)?",Rv(33)
2050 IF Y(0)<>0 THEN INPUT "Poisson's Ratio(0No,1Yes-long.strain on Y, 2Yes-trans.strain on X)?",Rv(34)
2060 IF (Rv(33)<=0) AND (Rv(34)<=0) THEN GOTO Read
2070 Rv(31)=Rv(33)
2080 IF Rv(33)=3 THEN Rv(31)=1
2090 Modulus:Rv(37)=.0002
2100 Rv(36)=2
2110 FOR J=1 TO C(8)-1
2120 IF (Pv(37)=G(J+1,Y(0))) AND (Pv(37)=G(J,X(0))) THEN GOTO 2140
2130 NEXT J
2140 Rv(37)=J
2150 IF C(1)=2 THEN Rv(37)=1
2160 PRINT "-----"
2170 IF Rv(34)/=1 THEN GOTO Poisson
2180 IF NOT (Rv(31)=2) THEN GOTO 2220
2190 PRINT "Secant Modulus."
2200 Y$="SECANT MODULUS, ksi"
2210 GOTO 2320
2220 PRINT "Tangent Modulus."
2230 Y$="TANGENT MODULUS, ksi"
2240 FOR J=Pv(36)+Pv(37) TO C(8)-Rv(36)
2250 Rv(29)=G(J+Rv(36),Y(0))-G(J-Rv(36),Y(0))
2260 Rv(30)=G(J+Rv(36),X(0))-G(J-Rv(36),X(0))
2270 Rv(8)=G(J-Pv(36)-Rv(37)+1,X(0))-G(J,X(0))
2280 Rv(8)=G(J-Rv(36)-Rv(37)+1,Y(0))-Rv(29)/Rv(30)
2290 NEXT J
2300 C(8)=C(8)-2*Rv(36)-Rv(37)+1
2310 GOTO 2410
2320 Rv(22)=G(Rv(37),X(0))
2330 Rv(23)=G(Rv(37),Y(0))
2340 FOR J=Rv(37)+1 TO C(8)
2350 Rv(29)=G(J,Y(0))-Rv(23)
2360 Rv(30)=G(J,X(0))-Rv(22)
2370 Rv(8)=G(J-Rv(37),X(0))-ABS(G(J,X(0)))
2380 Rv(8)=G(J-Rv(37),Y(0))-ABS(Rv(29)/Rv(30))
2390 NEXT J
2400 C(8)=C(8)-Rv(37)
2410 INPUT "Are axes to be plotted(0-1)?",Rv(28)
2420 INPUT "pen# to use(1-4)?",Rv(11)
2430 PEN Rv(11)
2440 PRINT "-----"
2450 IF Rv(28)=1 THEN GOTO Plot
2460 GOTO Plot1
2470 END
2480 Function:IF Pv(21)=0 THEN RETURN
2490 IF Pv(21)=1 THEN Rv(8)=Rv(8)/10*(1+

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2500 IF Rv(21)=1 THEN Rv(8)=ABS(Rv(8)/(C(2)-C(3))) $\times$ 1000
2510 IF Pv(21)=2 THEN Rv(8)=ABS(4+Rv(8)/(C(4)+C(9)))
2520 IF Rv(21)=3 THEN Pv(8)=ABS((-1.352994493E-5+1.352994493E1+Pv(8))/1.017)
2530 IF Rv(21)=4 THEN Rv(8)=ABS((-1.030318028E-4+1.386966577E1+Rv(8))/1.015)
2540 RETURN
2550 Maxmin: Rv(14)=Pv(12)=-1E99
2560 Rv(15)=Rv(13)=1E99
2570 FOR I=1 TO C(8)
2580 IF G(I,X(0))<Rv(13) THEN Rv(13)=G(I,X(0))
2590 IF G(I,X(0))>Rv(12) THEN Pv(12)=G(I,X(0))
2600 IF G(I,Y(0))<Pv(15) THEN Pv(15)=G(I,Y(0))
2610 IF G(I,Y(0))>Rv(14) THEN Rv(14)=G(I,Y(0))
2620 IF G(I,Y(0))/Rv(14) THEN Pv(32)=I
2630 NEXT I
2640 IF (C(1)=1) AND (Rv(33)=0) THEN C(8)=Pv(32)
2650 RETURN
2660 Poisson: PRINT "Poisson's Ratio."
2670 INPUT "X-axis: 1 long.strain, 2 trans.strain, 3 average of long.&trans.?"
,Rv(36)
2680 Y$="POISSON'S RATIO"
2690 FOR J=Rv(37) TO C(8)
2700 I=J-Rv(37)+1
2710 Rv(22)=G(J,X(0))
2720 Rv(23)=G(J,Y(0))
2730 IF Rv(34)=2 THEN Pv(22)=Rv(23)
2740 IF Rv(34)=2 THEN Pv(23)=G(I,X(0))
2750 G(I,Y(0))=Rv(23)/Rv(22)
2760 G(I,X(0))=Rv(22)
2770 IF Rv(36)=2 THEN G(I,X(0))=Pv(23)
2780 IF Rv(36)=3 THEN G(I,X(0))=(Rv(22)+Pv(23))/2
2790 NEXT J
2800 C(8)=I
2810 X$="LONGITUDINAL STRAIN, in/in."
2820 IF Rv(36)=2 THEN X$="TRANSVERSE STRAIN, in/in."
2830 IF Rv(36)=3 THEN X$="AVERAGE STRAIN(long.&trans.), in/in."
2840 GOTO 2410

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